

# Critical Reynolds number for onset of oscillatory flow around Ahmed body of slant angle 31 degree

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The Ahmed body is one of the most widely studied vehicle models in basic research on aerodynamic and drag-reduction for automobiles. Aerodynamic drag on the Ahmed body may change suddenly at slant angle  $\theta = 30^\circ$ , due to a transition of the flow<sup>\*1</sup>. This transition has been observed by the experiment for a Reynolds number of  $Re = \mathcal{O}(10^4)$  or higher, whereas no past numerical research has reproduced the transition of the flow. For such a high  $Re$ , direct numerical simulations require a huge number of computational grids to resolve the boundary layer flow and small vortices. Several components that are involved in the flow at a higher  $Re$ , such as trailing vortices and the separation-reattachment flow, possibly take place even for the low-Reynolds number conditions. In order to clarify this question, the authors' research group has carried out the series of three-dimensional numerical simulations. The critical Reynolds number for the slant angle  $\theta = 29^\circ$  and the mechanism of the oscillatory flow have been reported<sup>\*2</sup>

In the present study, a three-dimensional unsteady flow around the Ahmed body of the slant angle  $31^\circ$  for low Reynolds number is numerically investigated. The critical Reynolds number was calculated as  $Re_c = 819$ , and the Strohal numbers and the structure of the oscillatory flow were investigated. The amplitude of the oscillation was about half of that has been reported for  $29^\circ$ , whereas the oscillation was found to be extremely localized. At least the Reynolds number range for  $Re \leq 1000$ , the flow can be regarded as almost steady, because the oscillatory component is small and localized.

<sup>\*1</sup> S.R. Ahmed, G. Ramm, and G. Faltin, *SAE Paper 840300*, (1984).

<sup>\*2</sup> M. Mikasa *et al.*, *Trans. JSAE* **52**(6) p.1254 (2021).

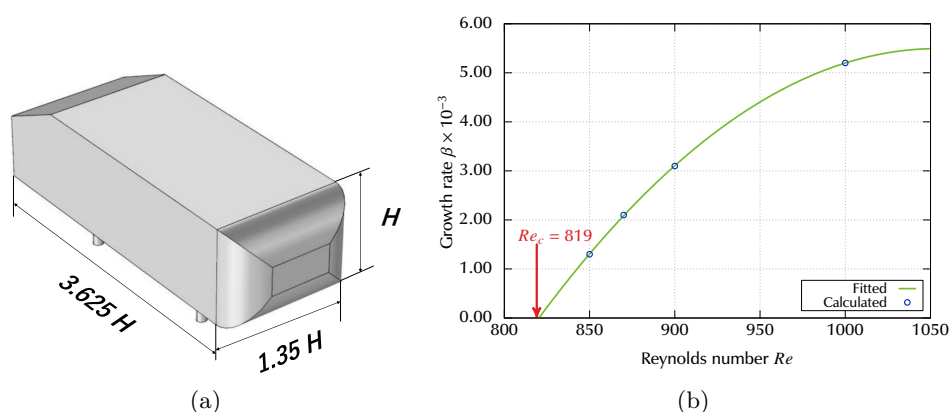


Fig.1: (a) Ahmed body. (b) Growth rate of the deviated flow component as a function of Reynolds number.

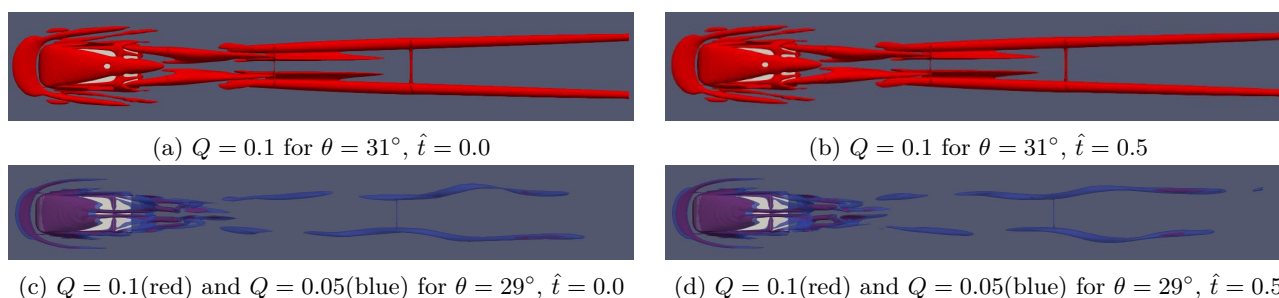


Fig.2: The contour surfaces of 2nd invariants of velocity gradient tensor  $Q$ .